Homework \#4 due: Tuesday, November 25

1. Recall the expression $W(z)=U\left(z+\frac{R^{2}}{z}\right)+\frac{\Gamma}{2 \pi i} \log z$ for potential flow past a cylinder with circulation. Consider the case $\frac{\Gamma}{2 \pi R}>2 U$, where $U>0, R>0, \Gamma>0$. In class it was stated that there is a single stagnation point in the interior of the flow in this case. Find the location of the stagnation point in terms of $U, R, \Gamma$. Plot some streamlines for the case $U=R=1, \Gamma=6 \pi$.
2. Recall the expression $W(z)=U \sqrt{z^{2}+a^{2}}$ for potential flow past a flat plate. The flow is unrealistic due to the absence of a wake, so consider a simple model in which the complex potential is modified by adding a counter-rotating pair of point vortices, one at $z=z_{0}$ with circulation $\Gamma$ and the other at $z=\overline{z_{0}}$ with circulation $-\Gamma$. Assume that $U>0$, real $z_{0}>0$.
a) Find the complex potential of the modified flow in the form $W_{1}(z)=W_{2}(\zeta(z)$ ), where $\zeta=z+\sqrt{z^{2}+a^{2}}$ is the conformal mapping from the exterior of a plate in the $z$-plane to the exterior of a cylinder in the $\zeta$-plane and $W_{2}(\zeta)$ is the modified complex potential in the $\zeta$-plane.
b) Find the circulation value $\Gamma^{*}$ satisfying the Kutta condition at the edges of the plate.
c) Plot some streamlines in the $z$-plane and $\zeta$-plane for the case $U=a=1, z_{0}=1+i, \Gamma=\Gamma^{*}$.
3. Let $\left(x_{j}(t), y_{j}(t)\right), j=1, \ldots, N$ denote a set of point vortices. Show that the following quantities are invariant in time.

$$
X=\sum_{j=1}^{N} \Gamma_{j} x_{j} \quad, \quad Y=\sum_{j=1}^{N} \Gamma_{j} y_{j} \quad, \quad R^{2}=\sum_{j=1}^{N} \Gamma_{j}\left(x_{j}^{2}+y_{j}^{2}\right)
$$

4. Consider two point vortices $z_{1}, z_{2}$ whose strengths satisfy $\Gamma_{1}>\Gamma_{2}>0$ and define the center of vorticity by $Z=\left(\Gamma_{1} z_{1}+\Gamma_{2} z_{2}\right) /\left(\Gamma_{1}+\Gamma_{2}\right)$. The previous exercise shows that $Z$ is invariant in time. Show that the point vortices $z_{1}, z_{2}$ travel on circles centered at $Z$, with the same angular velocity, but different radii $R_{1}, R_{2}$. What happens to $Z, R_{1}, R_{2}$ in the limit $\Gamma_{2} \rightarrow \Gamma_{1}$ ?
5. The stream function of a vortex-blob is $\psi_{\delta}(x, y)=-\frac{1}{2 \pi} \log \sqrt{x^{2}+y^{2}+\delta^{2}}$, where $\delta$ is a smoothing parameter. Note that $\psi_{0}$ is the stream function of a point vortex.
a) Find the vorticity of a vortex-blob, $\omega_{\delta}=-\Delta \psi_{\delta}$, and show that $\int_{\mathbf{R}^{2}} \omega_{\delta}(x, y) d x d y=1$ for all $\delta$. Plot $w_{\delta}(x, 0)$ for $-1 \leq x \leq 1$ with $\delta=0.2,0.1,0.05$ (all on the same plot).
b) The evolution equations for a set of vortex-blobs are the same as for a set of point vortices except that $\psi_{\delta}$ is used instead of $\psi_{0}$. Compute the motion of a set of vortex-blobs having initial locations $x_{j}(0)=\cos \theta_{j}, y_{j}(0)=0, \theta_{j}=\left(-1+\frac{j}{N+1}\right) \pi$ and strengths $\Gamma_{j}=\frac{\pi}{N+1} \cos \theta_{j}$, for $j=1, \ldots, N$. Take $N=200, \delta=2 \cdot 10^{-1}$ and use any convenient time integration scheme (code it yourself or use Matlab). Plot the blob locations at $t=0,1,2,4,8,16$ (use axis square in Matlab for proper scaling). The results represent a 2D model for the wake behind an airplane; see pages 50-51 in Van Dyke's photo album. Accounting for 3D effects is an important ongoing research problem.
